1. Introduction

The IAEA Regulations for the Safe Transport of Radioactive Material [1] in its paragraph 621 require that all packages of radioactive material (excepted packages, industrial packages IP-1, IP-2 and IP-3, type A, type B(U), type B(M) and type C packages) be designed to meet the following requirement:

- “Packages containing radioactive material to be transported by air shall be capable of withstanding, without loss or dispersal of radioactive contents from the containment system, an internal pressure that produces a pressure differential of not less than the maximum normal operating pressure plus 95 kPa.”

This requirement may be difficult to achieve, especially in the case of packagings of large dimensions, such as ISO freight containers which are commonly used as type IP-2 packages for air carriage of tools and components as surface contaminated objects (SCO-I or SCO-II), and it may seem excessively severe for low activity materials with little risk even in case of dispersion, such as excepted packages loaded with solid radioactive materials for example.

A first proposal was submitted by WNTI in the 2013 review cycle [2].

As recommended during the “Technical Meeting (TM) on the Environment to which Packages may be subject during Transport and related Issues Concerning the Regulations for the Safe Transport of Radioactive Material” (IAEA TM-44891) which took place from 15 to 19 July 2013 [3], the WNTI submitted for comment this proposal to the Dangerous Goods Panel of ICAO [4] during its 24th meeting (28 October to 8 November 2013).

The comment made by the Panel was the following [5]:

- “While some panel members understood the logic in developing different requirements based on the level of risk in that this approach was applied to other dangerous goods, they believed that breaking the requirements into three different categories would complicate the provisions and make them more difficult to understand. They did not
feel there was sufficient justification for lowering the standards and that the current provisions should remain.”

A second proposal was submitted by WNTI during the 2015 review cycle [6]. Compared to the 2013 proposal, it was supported by more accurate justifications, and the final proposal was simplified. This proposal was revised during the review cycle to take account the comments provided by participants during the IAEA TRANSSC Extraordinary meeting in September 2015 [7].

The resulting final proposal as submitted to the ordinary TRANSSC meeting of November 2015 is presented in Appendix 1.

The representative of ICAO in the TRANSSC indicated that this proposal could not be accepted as proposed and requested that at least one further technical meeting between experts should be held prior to its acceptance by TRANSSC. As a result, this final proposal could not be accepted for the current 2015 review cycle.

The TRANSSC decision as recorded was the following (see proposal WNTI/2015/03 Rev.2 in [8]):

- “No action taken. Decision on the proposal is deferred to the IAG (Inter Agencies Group) and other appropriate aviation organizations for further discussion and development.”

2. Objective

The objective of this technical document is to provide necessary information as required to initiate technical exchange on this issue between experts, as recommended by the TRANSSC.

3. Intent of the proposal for change

This proposal was intended to adapt requirements regarding the pressure differential, which packages containing radioactive material shall be able to withstand without release or dispersion of radioactive material, to the levels of risk presented by real contents of the packages, with regard to safety of the aircraft, passengers and crew members.

It concerns mainly package designs not subject to approval and carrying solid substances or objects, while maintaining the basic safety objective: packages of radioactive material shall retain their radioactive contents in situations of variation of ambient pressure around packages as generated during transport by air.
4. Current regulatory requirements regarding resistance to pressure of packages for transport of dangerous goods

The requirements summarized hereafter are those from the Regulations for the Safe Transport of Radioactive Material of IAEA [1] and from the Technical Instructions for the Safe Transport of Dangerous Goods by Air [9].

a. Radioactive materials:

Para 620 of IAEA SSR-6; para 7.2.2 of part 6 of ICAO-TI:

Packages to be transported by air shall be so designed that if they were exposed to an ambient temperature ranging from -40°C to +55°C, the integrity of containment would not be impaired.

Para 621 of IAEA SSR-6; para 7.2.3 of part 6 of ICAO-TI:

Packages containing radioactive material to be transported by air shall be capable of withstanding, without loss or dispersal of radioactive contents from the containment system, an internal pressure that produces a pressure differential of not less than the maximum normal operating pressure plus 95 kPa.

Para 229 of IAEA SSR-6; para 3.1 of part 1 of ICAO-TI:

Maximum normal operating pressure (MNOP): For the transport of radioactive materials, shall mean the maximum pressure above atmospheric pressure at mean sea level that would develop in the containment system in a period of one year under the conditions of temperature and solar radiation corresponding to environmental conditions in the absence of venting, external cooling by an auxiliary system or operational controls during transport.

b. Infectious substances: (packing instructions 620 and 650 of ICAO-TI)

The primary receptacle or the secondary packaging shall be capable of withstanding without leakage, an internal pressure of 95 kPa (0.95 bar).

c. Liquid dangerous goods: (paras 1.1.6 and 1.1.16 of part 4, and para 4.5.3 of part 6 of ICAO-TI)

Packagings for which retention of liquid is a basic function shall be capable of withstanding, without leakage, a rated internal pressure as shown below:

As required in para 1.1.6 of part 4 of ICAO-TI (valid for inner packagings of combination packagings):

(a). \( PT = (P_{M55} \times 1.5) \text{ kPa with a minimum of } 95 \text{ kPa or } 75 \text{ kPa (liquids in packing group III of class 3 or division 6.1)} \); or

(b). \( PT = (V_{p50} \times 1.75) - 100 \text{ kPa with a minimum of } 95 \text{ kPa}; \) or

(c). \( PT = (V_{p55} \times 1.5) - 100 \text{ kPa with a minimum of } 95 \text{ kPa}. \)
As required in para 1.1.16 of part 4 and para 4.5.3 of part 6 of ICAO-TI (valid for packagings other than inner packagings of a combination packaging):

(a). \[ PT = (P_{M55} \times 1.5) \text{ kPa with a minimum of 75 kPa (liquids in packing group III of class 3 or division 6.1), 250 kPa (liquids in packing group I), or 95 kPa (other liquids); or} \]

(b). \[ PT = (V_{P50} \times 1.75) - 100 \text{ kPa with a minimum of 250 kPa (liquids in packing group I), or 100 kPa (other liquids); or} \]

(c). \[ PT = (V_{P55} \times 1.5) - 100 \text{ kPa with a minimum of 250 kPa (liquids in packing group I), or 100 kPa (other liquids).} \]

Where

\[ PT = \text{test pressure in kPa (gauge);} \]

\[ P_{M55} = \text{pressure measured in the filled packaging at the temperature of 55°C (with filling temperature of 15°C);} \]

\[ V_{P50} = \text{vapour pressure at 50°C;} \]

\[ V_{P55} = \text{vapour pressure at 55°C.} \]

5. Basic examples for certain materials illustrating the use of the current regulatory requirements regarding resistance to the pressure of packagings for transport of dangerous goods

The examples developed are based on the following materials:

- A solid material (either radioactive or presenting another danger);
- Liquid having properties of water (either radioactive material or presenting another danger);
- Liquid having properties of ethanol (either pure – Class 3, PG-II, or contaminated by radioactive material).

The following assumptions are considered:

- Materials do not produce heat, and do not emit gases by radiolysis or thermolysis;
- Packagings are filled and closed at 15°C, at an ambient pressure of 100 kPa (the air + vapour pressure of the material at 15°C);
- The degree of filling for liquids is that prescribed in ADR [10] in para 4.1.1.4 (a) (according to the boiling temperature), degree of filling that allows to check that the packaging is not completely filled at 55°C as required in § 1.1.5 of part 4 of the ICAO-TI;
- The maximum ambient temperature of 55°C is reached.
The following calculations are made:

- Total pressure (absolute) in the package at 55 °C (P55), taking into account pressure increase due to:
  - The temperature increase of gases imprisoned at the closure of the packaging (increase from 15°C to 55°C);
  - The expansion of the liquids between 15°C and 55°C;
  - The increase of vapour pressure of the liquids between 15°C and 55°C;
- Maximum normal operating pressure (MNOP): gauge pressure developed in the packaging at 55°C (PM55)
  - \( PM_{55} = P_{55} - 100; \)

**a. Solid material or object loaded in a packaging at 15°C and 100 kPa:**

The only pressure generated is due to temperature increase of the gases imprisoned at closure, there being no dilatation (expansion) of the solids between 15°C and 55°C, and no vapour or gas produced:

\[
PM_{55} = P_{15} \times \frac{(55 - 15)}{(273 + 15)} = 100 \times \frac{(55 - 15)}{(273 + 15)} = 13.89 \text{ kPa} \approx 14 \text{ kPa}
\]

MNOP = \( PM_{55} = 14 \text{ kPa} \)

PT for Radioactive material: MNOP + 95 kPa = 14 + 95 = 109 kPa = **110 kPa**

PT for solid infectious substances: **95 kPa**

PT for other solid dangerous goods: **Not applicable**

**b. Water loaded in a packaging at 15°C and 100 kPa:**

Physical data for water: see [11]

Degree of filling at 15°C (see 4.1.1.4 of ADR [10]): 94%

Density of water at 15°C: 0.999

Density of water at 55°C: 0.986

Degree of filling at 55°C: \((0.999/0.986) \times 0.94 = 0.9524\)

Vapour pressure of water at 15°C: 1.705 kPa;

Vapour pressure of water at 55°C: 15.760 kPa;

Air pressure at 15°C imprisoned in the package: 100 – 1.705 = 98.295 kPa
Pressure increase of the air imprisoned between 15°C and 55°C due to liquid expansion and increase in temperature:

\[
98.295 \times \frac{(1 - 0.94)}{(1 - 0.9524)} \times \frac{(273 + 55)}{(273 + 15)} - 98.295 = 42.815 \text{ kPa}
\]

Water vapour pressure increase between 15°C and 55°C: 15.760 - 1.705 = 14.055 kPa

Total pressure increase between 15°C and 55°C: \(P_{M55} = 42.815 + 14.055 = 56.87 \text{ kPa}\)

\[MNOP = P_{M55} = 56.87 \text{ kPa rounded to 57 kPa}\]

PT for Radioactive material: \(MNOP + 95 \text{ kPa} = 57 + 95 = 152 \text{ kPa} = 155 \text{ kPa}\)

PT for liquid dangerous goods:

(a) \(PT = (P_{M55} \times 1.5) \text{ kPa} = 57 \times 1.5 = 85.5 \text{ kPa rounded to 90 kPa}\)

\[=> \text{ minimum PT} = 90 \text{ kPa , 95 kPa or 250 kPa (see following table)}\]

(b) \(PT = (Vp_{50} \times 1.75) - 100 \text{ kPa} = (12.352 \times 1.75) - 100 = -78.384\)

\[=> \text{ minimum PT} = 95 \text{ kPa , 100 kPa or 250 kPa (see following table)}\]

(c) \(PT = (Vp_{55} \times 1.5) - 100 \text{ kPa} = (15.760 \times 1.5) - 100 = -76.360\)

\[=> \text{ minimum PT} = 95 \text{ kPa , 100 kPa or 250 kPa (see following table)}\]

<table>
<thead>
<tr>
<th>Nature of the liquid: water</th>
<th>Inner packaging of a combination packaging</th>
<th>Single packaging</th>
</tr>
</thead>
<tbody>
<tr>
<td>Class 7</td>
<td>155 kPa</td>
<td>155 kPa</td>
</tr>
<tr>
<td>PG III class 3 or division 6.1</td>
<td>(a): 90 kPa</td>
<td>(a): 90 kPa</td>
</tr>
<tr>
<td></td>
<td>(b): 95 kPa</td>
<td>(b): 100 kPa</td>
</tr>
<tr>
<td></td>
<td>(c): 95 kPa</td>
<td>(c): 100 kPa</td>
</tr>
<tr>
<td>PG III other classes or divisions + PG-II</td>
<td>(a): 95 kPa</td>
<td>(a): 95 kPa</td>
</tr>
<tr>
<td></td>
<td>(b): 95 kPa</td>
<td>(b): 100 kPa</td>
</tr>
<tr>
<td></td>
<td>(c): 95 kPa</td>
<td>(c): 100 kPa</td>
</tr>
<tr>
<td>PG I</td>
<td>(a): 95 kPa</td>
<td>(a): 250 kPa</td>
</tr>
<tr>
<td></td>
<td>(b): 95 kPa</td>
<td>(b): 250 kPa</td>
</tr>
<tr>
<td></td>
<td>(c): 95 kPa</td>
<td>(c): 250 kPa</td>
</tr>
</tbody>
</table>
c. Ethanol loaded in a packaging at 15°C and 100 kPa:

Physical data for ethanol: see [12]

Degree of filling at 15°C (see 4.1.1.4 of ADR [10]): 92%

Density of ethanol at 15°C: 0.795
Density of ethanol at 55°C: 0.758

Degree of filling at 55°C: $(0.795/0.768) \times 0.92 = 0.9523$

Vapour pressure of ethanol at 15°C: 4.339 kPa;
Vapour pressure of ethanol at 55°C: 37.459 kPa;

Air pressure imprisoned in the package at 15°C: $100 - 4.339 = 95.661$ kPa

Pressure increase of the air imprisoned between 15°C and 55°C due to liquid expansion and increase in temperature:

$$95.661 \times \frac{(1 - 0.92)}{(1 - 0.9523)} \times \frac{(273 + 55)}{(273 + 15)} - 95.661 = 87.06 \text{ kPa}$$

Ethanol vapour pressure increase between 15°C and 55°C:

$$37.459 - 4.339 = 33.120 \text{ kPa}$$

Total pressure increase between 15°C and 55°C: $P_{M55} = 87.06 + 33.12 = 120.18 \text{ kPa}$

$MNOP = P_{M55} = 120.18 \text{ kPa}$ rounded to 121 kPa

PT for Radioactive material: $MNOP + 95 \text{ kPa} = 121 + 95 = 216 \text{ kPa} \approx 220 \text{ kPa}$

PT for liquid dangerous goods:

(a) $PT = (P_{M55} \times 1.5) \text{ kPa} = 121 \times 1.5 = 181.5 \text{ kPa}$ rounded to $185 \text{ kPa}$

=> minimum PT = $185 \text{ kPa}$, (see following table)

(b) $PT = (Vp_{50} \times 1.75) - 100 \text{ kPa} = (29.598 \times 1.75) - 100 = - 48.204$

=> minimum PT = $95 \text{ kPa}$ or $100 \text{ kPa}$ (see following table)

(c) $PT = (Vp_{55} \times 1.5) - 100 \text{ kPa} = (37.459 \times 1.5) - 100 = - 43.812$

=> minimum PT = $95 \text{ kPa}$ or $100 \text{ kPa}$ (see following table)
6. Conditions of pressure and pressure variation that may be encountered during air transportation

a. Maximum and minimum ambient pressure in aircraft cargo holds

The transport of goods by air is carried out either by means of:

- pressurized aircrafts (that is passenger aircrafts and cargo aircrafts adapted from passenger aircraft designs, such as Airbus and Boeing aircrafts), or,
- non-pressurized or partially pressurized aircrafts (usually aircrafts specifically designed for the transport of heavy and/or bulky cargo, such as the Antonov-124).

For reasons related to health and comfort of passengers and crew members, the cabin altitude (equivalent pressure) is limited in commercial pressurized aircrafts to 8,000 feet (about 2,500 m) corresponding to an ambient pressure of 75 kPa [13] [14]. Above this altitude, without additional oxygen in the air to breathe, there is a risk of hypoxia (hypoxia is a decrease in available oxygen below the amount required for normal functioning of the human body). There is no pressure differential between the cabin and the cargo compartments in these aircrafts.

For reasons related to safety of crew members, the cruising altitude of non-pressurized aircrafts, or cabin altitude in the cargo compartments of partially pressurized aircrafts, is maintained below 34,000 feet (10360 m) (the partial pressure of oxygen in the lungs at 34,000 feet altitude is the same as that for a person breathing air at sea level, 34,000 feet is therefore the highest altitude at which a person would be provided complete protection from the effects of hypoxia [15]). This altitude corresponds to an atmospheric pressure of 25 kPa. For the health and comfort of the crew, either they are supplied with oxygen, or the aircraft has a pressurized cabin separated from the cargo bay, as is the case of the Antonov-124.

Thus, as indicated in note 3 of the preliminary notes to part 4 of the ICAO TI [9], under routine transport conditions, the ambient pressure around packages transported in an aircraft may vary:

- from 75 kPa to 100 kPa in a pressurized aircraft, and,
- from 25 kPa to 100 kPa for a non-pressurized or partially pressurized aircraft.

This note 3 was incorporated in the 2011/2012 Edition the ICAO TI on the base of a working paper related to the “transportation of dangerous goods in non-pressurized cargo holds and pressure differentials in flight”, presented at a Dangerous Goods Panel of ICAO in May 2009 [16].
b. Maximum pressure rate variations in cargo holds of aircrafts under routine conditions of transport

Climb and descent phases of an aircraft induce variation of pressure inside the aircraft (decrease from 100 kPa to 75 kPa in the climb phase or rise from 75 kPa to 100 kPa in the descent phase of a pressurized aircraft, decrease from 100 kPa to 25 kPa in the climb phase or rise from 25 kPa to 100 kPa in the descent phase of a non-pressurized or partially pressurized aircraft).

In a non-pressurized or partially pressurized aircraft, the variation of pressure inside the cargo bay is close to the change in air pressure around the aircraft corresponding to its rate of climb or descent. The rate of climb of commercial aircraft in normal operation is of the order of 2,500 feet per minute, which corresponds to a decrease in pressure of 150 Pa/s or 9 kPa per minute [17].

In pressurized aircraft, for the comfort of passengers and crew members (in particular to avoid barotrauma – ear pain), the rate of climb in cabin altitude is limited to 500 feet per minute which corresponds to a pressure decrease of 30 Pa/s or 1.8 kPa per minute, and the rate of descent is limited to 300 feet per minute, which corresponds to a pressure increase of 18 Pa/s or 1.1 kPa per minute [18] [19].

In normal situations of operation, which could be considered as routine conditions of transport, packages are then subject to a maximum variation of ambient pressure of 1.8 kPa per minute or 9 kPa per minute depending on whether transported in a pressurized aircraft or a non-pressurized or partially pressurized aircraft.

Those conditions are summarized in Figure 1.

Figure 1:
Typical altitudes, pressures and variations in an aircraft in normal operation

- Typical altitudes
c. Pressure rate variations in cargo holds of aircrafts in case of abnormal depressurization (rapid or explosive depressurization)

Rapid depressurization (also known as "explosive depressurization" when the rate of climb in cabin altitude exceeds 7000 feet per minute) [19], is usually the result of a hardware failure of the aircraft (dysfunction of the pressurization system, inadequate closure of a door, failure of a seal leading to leakage, cracking or loss of a window ...). This phenomenon only really concerns pressurized aircraft. It leads to triggering of alarms and engagement of an emergency procedure by the crew, consisting in particular in descending of the plane as rapidly as possible to an altitude such that crew members and any passengers find normal breathing conditions (at 10,000 feet or less).

These are situations that occur fairly regularly (such events occur every year, only the most spectacular cases being relayed in the media) for which usually the aircraft, passengers and crew survive, and often without major injury. For example, for a total of 517 events involving Australian aircraft in the period from January 1975 to March 2006, only one resulted in the loss of the aircraft and fatal injuries to occupants and only 4 led to serious injury to occupants according to a specific study by the Australian Aviation Authority [19].

In France, such events also occur, subject to analysis of the BEA (BEA is the authority responsible for safety investigations for civil aviation accidents in France). From reports established by the BEA, an illustrative example of what may be considered a serious depressurization event is that which occurred on November 21, 2007 over the South of France to an Airbus A330 at an altitude of 41,000 feet (12 500 m) during a demonstration flight prior to delivery to an airline [20]. Tests were related to pressurization systems. Malfunction of the valve command system occurred, leading to a sudden reduction in cabin pressure. The cabin altitude increased from 7,500 feet to 30,500 feet in a little more than 2 minutes, corresponding to a rate of climb of over 10,000 feet per minute (a decrease of pressure from 76.7 kPa to 29.4 kPa) according to the graph of the cabin altitude records. These values rank this event amongst those considered as explosive depressurizations. Two people who were not immediately fitted with oxygen masks fainted, several people suffered from barotrauma (ear pain), but finally no serious injuries occurred. The flight was stabilized at 10,000 feet (3050 m, 69.5 kPa) a little more than 5 minutes after the beginning of the incident. The aircraft suffered no damage.

In the Australian study mentioned above, no event, among the 39 for which data on cabin pressure were available reached this cabin depressurization rate (the maximum recorded in this study was 6500 feet per minute) [19]. The depressurization event suffered by the Airbus A330 over France may be considered as being among the more severe events. It is therefore proposed to consider it as a reference incident to represent the conditions of depressurization that may be experienced by a package or container carried on an aircraft during such an event. The reference event is represented in figure 2 below.

This event confirms, as does the ASTB report, that any people involved may survive, without serious injury, a pressure reduction rate of almost 0.5 bar (50 kPa) in 2 minutes, following a depressurization incident at 12 500 m altitude where the ambient pressure is 17.77 kPa. It also reveals that the minimum pressure in the cabin (29.4 kPa) after the event is not necessarily that prevailing outside the aircraft at the start of the event, and that it remains higher than the minimum ambient pressure indicated in note 3 of the preliminary notes to part 4 of the ICAO TI (25 kPa).
It also reveals the need for large volume containers or packagings to either:
- be fitted with quick-opening pressure-equalization systems in order to balance pressure without risk of bursting of the container (which could adversely affect the safety of persons on board and endanger the safety of the aircraft), or,
- to be designed to withstand the pressure differential generated by this kind of situation. An ISO standard specifically addresses this issue [17]. This standard applies to airtight shipping containers with an internal volume of 1 m³ (35 ft³) or more.

The question raised during TRANSSC 31, and to which experts shall answer is as follows:

**Shall rapid and explosive depressurizations be considered as part of the “normal conditions of transport” as defined by the IAEA?**

**Figure 2:**
Typical altitudes, pressures and variations in an aircraft in a severe depressurization incident

- **Typical depressurization incident (A330 – Toulouse 2006)**

Other data that may be used for defining of conditions to be considered as representative of normal conditions of transport:

Certification standards applicable to large aircrafts [13] [14] (see Appendix 2) require that:

- If certification for operation above 25,000 feet is required, the aircraft must be designed so that occupants will not be exposed to cabin pressure altitudes in excess of 15,000 feet after any probable failure condition in the pressurization system.
- Failures or a combination of failures which expose occupants to: (1) cabin altitudes in excess of either 25,000 feet for more than 2 minutes, or (2) cabin altitudes that exceed 40,000 feet for any duration, shall be shown to be extremely improbable.
- The absolute pressure at 15,000 feet is 57.2 kPa, at 25,000 feet is 37.65 kPa and at 40,000 feet is 18.7 kPa.
7. Comparison between the performance required for packagings and pressure variations that may be encountered during air transport

The table in the following page summarizes various situations that are covered by the requirements for packages against ambient pressure variations around packages during air transport (decrease of 25 kPa for pressurized aircraft, 75 kPa for non-pressurized or partially pressurized aircraft) when considering increase of ambient temperature from 15°C to 55°C for the three examples of material for which the required pressure test (PT) has been determined in paragraph 5 above (solid, water and ethanol).

Under normal flight conditions (routine conditions of transport) requirements regarding the pressure resistance of packagings containing liquid dangerous goods (other than class 7), and what they shall be able to withstand without leakage, may be unsufficient for situations where the maximum increase of ambient temperature (15°C to 55°C) and the maximum decrease of ambient pressure around the packages (100 kPa to 25 kPa) occur simultaneously (this is verified for water PG-II or III in non-pressurized aircraft and for ethanol when formulas (a), (b) or (c) are used).

The fact that the formulae provided in paras 1.1.6 and 1.1.16 of part 4, and para 4.5.3 of part 6 of ICAO-TI [9] are still considered as satisfactory shows that combination of the highest decrease in ambient pressure with the highest increase in ambient temperature around the package is certainly of low or very low probability.

These examples also highlight that the formulas provided in (a), (b) and (c) are somewhat "approximate":

- in not considering directly the reduction of ambient pressure in the formulae (a), (b) and (c), situations may occur that, even with account taken of the minimum test pressure, the final pressure specified is not sufficient (this is the case for water);
- by neglecting the effect of imprisoned air at closing of the packagings, formulae (b) and (c) may lead to serious underestimation of the minimum pressure the packaging used shall be able to withstand, especially in case of liquid with low vapour pressure at 15°C (this is the case for ethanol).

As for the rules specified for Class 7 transports, these examples show that the margin is always greater than 20 kPa.

The rule applicable for Class 6.2 shows that minimum pressure of 95 kPa in case of solid material is sufficient against all situations (the margin is always greater than 6 kPa).

The proposal for Class 7 to replace the "internal pressure that produces a pressure differential of not less than the maximum normal operating pressure plus 95 kPa" by “the maximum gauge pressure that may be developed in the package at 55°C when it has been filled at 15°C (or the maximum normal operating pressure) plus 75kPa, with a minimum of 95kPa” was based on these findings.

This proposal covers all situations with a positive margin. The minimum total pressure difference of 95 kPa allows to cover the case of solid (non-heating, without gas production) packed with air at an ambient temperature of 15°C at sea level, without need for any supplementary justification.
<table>
<thead>
<tr>
<th>Material</th>
<th>PM55</th>
<th>Class</th>
<th>PT</th>
<th>$P_{\text{ambient decrease}} , ^\ast$</th>
<th>$\Delta P , \text{maxi}$</th>
<th>Margin</th>
</tr>
</thead>
<tbody>
<tr>
<td>Solid</td>
<td>14 kPa</td>
<td>7</td>
<td>110 kPa</td>
<td>-25 kPa</td>
<td>39 kPa</td>
<td>+71 kPa</td>
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<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>89 kPa</td>
<td>+21 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td>6.2</td>
<td>95 kPa</td>
<td>-25 kPa</td>
<td>39 kPa</td>
<td>+56 kPa</td>
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<td>-75 kPa</td>
<td>89 kPa</td>
<td>+6 kPa</td>
</tr>
<tr>
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<td></td>
<td>Other than 6.2 and 7</td>
<td>Non required</td>
<td>-25 kPa</td>
<td>39 kPa</td>
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</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>89 kPa</td>
<td>-89 kPa</td>
</tr>
<tr>
<td>Liquid</td>
<td>57 kPa</td>
<td>7</td>
<td>155 kPa</td>
<td>-25 kPa</td>
<td>82 kPa</td>
<td>+73 kPa</td>
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<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>132 kPa</td>
<td>+23 kPa</td>
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<td></td>
<td></td>
<td>3, PGIII 6.1, PGIII All packagings</td>
<td>90 kPa (a)</td>
<td>-25 kPa</td>
<td>82 kPa</td>
<td>+8 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>132 kPa</td>
<td>-42 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td>All PGI, all PGII and other PGIII Inner packaging</td>
<td>95 kPa (a) (b) (c)</td>
<td>-25 kPa</td>
<td>82 kPa</td>
<td>+13 kPa</td>
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<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>132 kPa</td>
<td>-37 kPa</td>
</tr>
<tr>
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<td></td>
<td>3, PGIII 6.1, PGIII Inner packaging</td>
<td>95 kPa (b) (c)</td>
<td>-25 kPa</td>
<td>82 kPa</td>
<td>+13 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>132 kPa</td>
<td>-37 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td>All PGII and all PGIII Single packaging</td>
<td>100 kPa (b) (c)</td>
<td>-25 kPa</td>
<td>82 kPa</td>
<td>+18 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
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<td>-75 kPa</td>
<td>132 kPa</td>
<td>-32 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td>All PG I Single packaging</td>
<td>250 kPa (a) (b) (c)</td>
<td>-25 kPa</td>
<td>82 kPa</td>
<td>+168 kPa</td>
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<td></td>
<td>-75 kPa</td>
<td>132 kPa</td>
<td>+118 kPa</td>
</tr>
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<td>Liquid</td>
<td>121 kPa</td>
<td>7</td>
<td>220 kPa</td>
<td>-25 kPa</td>
<td>146 kPa</td>
<td>+74 kPa</td>
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<td></td>
<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>196 kPa</td>
<td>+24 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td>PG II class 3 all packagings</td>
<td>185 kPa (a)</td>
<td>-25 kPa</td>
<td>146 kPa</td>
<td>+39 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>196 kPa</td>
<td>-11 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td>PG II class 3 inner packaging</td>
<td>95 kPa (b) (c)</td>
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<td>146 kPa</td>
<td>-51 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>196 kPa</td>
<td>-101 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td>PG II class 3 single packaging</td>
<td>100 kPa (b) (c)</td>
<td>-25 kPa</td>
<td>146 kPa</td>
<td>-46 kPa</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td></td>
<td>-75 kPa</td>
<td>196 kPa</td>
<td>-96 kPa</td>
</tr>
</tbody>
</table>

* -25 kPa in case of pressurized aircraft, -75 kPa in case of non-pressurized aircraft
There is no requirement for minimum pressure resistance for packagings containing solid dangerous goods other than Class 7 and Class 6.2 in the ICAO-TI [9]. However, with the exception of Class 7, the maximum net quantity of solid dangerous goods per package is generally limited to 200 kg or less, with some exceptions of up to 400 kg as for UN3335, which means that the corresponding packagings are of size less than 1 m³.

Packagings used for radioactive materials are not submitted to such mass or volume limitations. In particular, freight containers may be used as IP-2 or IP-3 package, as permitted by para 629 of IAEA SSR-6 [1] and by para 7.4.4.2 of Part 6 of ICAO-TI [9]. Such freight containers are much larger than 1 m³, and are clearly not designed to sustain significant differential pressure (at most a few kPa).

For containers larger than 1 m³, carried by air, ISO standards applicable to the design and testing of air freight containers contain requirements to avoid dangerous overpressure of the containers under normal flight conditions [17] [21] which could endanger persons on board or the aircraft itself in the event of break-up: so either the container is designed to sustain such a differential pressure or the container is equipped with valves or venting allowing equalization of inner and external pressure during climbing and descent phases of the flight with a maximum climb rate of 2,500 ft/minute (150 Pa/s = 9 kPa/minute).

The proposal for an alternative requirement for solid material or objects of low radioactive hazard ((a) packages containing only special form radioactive material, SCO-I or SCO-II; (b) excepted packages, industrial packages and type A packages, containing solid radioactive material, excluding powders; (c) packages containing LSA-I material in powder form) takes account of these facts, with a minimum requirement to guarantee an equivalent level of safety: The packages shall be able to prevent loss or dispersal of their radioactive contents with a reduction of ambient pressure to 25 kPa. The necessary recommendations and guidance for filters and valves was provided in para 621.2 of the proposal (see appendix 1). In particular the following conditions are considered to represent “routine conditions of transport”:

- temperature of filling: 15°C, maximum temperature during transport: 55°C
- maximum decrease of ambient pressure in cargo holds of an aircraft: 75 kPa;
- minimum ambient pressure in cargo holds of an aircraft: 25 kPa;
- maximum pressure rate decrease during climb phase: 9 kPa/minute (non-pressurized or partially pressurized aircraft) or 1.8 kPa/minute (pressurized aircraft).

In the case of depressurization incident as illustrated in Figure 2, the maximum resulting decrease in ambient pressure remains equal to 75 kPa. The results in the table presented in paragraph 7 (a) above, show that, except for packagings for Class 7 and Class 6.2 (solid), most packagings used for liquids will not be able to support such a pressure differential when combined with the maximum increase of the ambient temperature.

The question which may arise is:

Taking into account the fact that most packagings authorized for the transport of liquid dangerous goods by air are unable to withstand, without leakage, the rapid and explosive depressurization of a pressurized aircraft, is it then justified to consider that rapid and explosive depressurization should nevertheless be considered as representing “normal conditions of transport” for all types of package containing radioactive material?
For air freight containers larger than one cubic meter, standards applicable to the design and testing of these air freight containers contain requirements to avoid explosion of these containers in case of emergency situations following a depressurization incident occurring at cruise altitude [17] [21]: either the container is designed to sustain such a pressure differential without explosion, or the container is equipped with emergency valves or vents as required in ISO standard [17] to avoid any over-pressurization of the container in these emergency situations.

For packagings with equipment permitting equilibrium of pressure during the climbing phase, the figure 2 and the report [19] show that the maximum pressure variation during a depressurization event at cruise altitude may be considered as being less than 50 kPa:

- Either the packaging shall be able to sustain such a pressure differential without explosion,
- or the packaging shall be equipped with emergency valves or vents as required in ISO standard [17] to avoid any over-pressurization of the package in these emergency situations.

8. Graded approach

The IAEA SSR-6 [1] considers a graded approach which may be summarized as shown in Figure 3 below:

**Figure 3:**
Graded approach used in IAEA SSR-6
(Requirements adapted to the risk level presented by the package content)
The proposal for review of pressure differential requirements applicable to packages containing radioactive material transported by air is also based on a graded approach, taking into account the level of danger of the content and its potential to escape from the package in case of leakage, and that depressurization incidents should be considered as normal conditions of transport:

- withstand a pressure differential of at least MNOP + 75 kPa with a minimum of 95 kPa without loss or dispersal of the radioactive contents
  - B(U), B(M) and C
  - IF, AF, B(U)F, B(M)F and CF
  - H(U), H(M)
  - Liquid LSA-I, liquid LSA-II, LSA-II powders,
  - liquid or gas in excepted package or in Type A

- prevent loss or dispersal of the radioactive contents with a reduction of ambient pressure to 25kPa (pressure rate variation of 9 kPa/minute), including in case of depressurization incident (drop of the ambient pressure from 75 to 25 kPa in less than 2 minutes)
  - Special form in Type A, B(U), B(M) or C
  - LSA-III, solid LSA-II excluding powders, SCO-II, solid material in a Type A

- prevent loss or dispersal of the radioactive contents with a reduction of ambient pressure to 25kPa (pressure rate variation of 9 kPa/minute)
  - Solid LSA-I, SCO-I, solid material in an excepted package

The updated proposal taking account the fact that depressurization incidents should be considered as part of normal conditions of transport is in Appendix 3.

9. References

[7] IAEA – proposal WNTI/2015/03 (rev.2) (also referenced INF-05bWNTI201503Rev 2)
[8] IAEA - TRANSSC 31 Decisions Record (Rev.1)
[10] European Agreement Concerning the International Carriage of Dangerous by Road (ADR) – applicable as from 1st January 2015:
[12] Physical data of ethanol – CELSIUS
[14] EU EASA CS25 Section 25.841
[16] ICAO DGP-WG/09-WP/46 – Transportation of dangerous goods in non-pressurized cargo holds - Auckland, New Zealand, 4 to 8 May 2009
[18] US FAA Aviation Maintenance Technician Handbook – Airframe - Chapter 16 - Control of Cabin Pressure - Pressurization Modes
Appendix 1

Proposed amendment of the para 621 of IAEA SSR-6 and of the para 621.1 and following of IAEA SSG-26 submitted by WNTI in 2015

The final proposal of WNTI as it was presented at the 31st regular meeting of the IAEA TRANSSC in November 2015 was:

621. Packages containing radioactive material to be transported by air shall be capable of withstanding, without loss or dispersal of radioactive contents from the containment system, an internal pressure that produces a pressure differential of not less than the maximal gauge pressure that may be developed in the package at 55°C when it has been filled at 15°C (or the maximum normal operating pressure) plus 75kPa, with a minimum of 95kPa. This requirement is not applicable to the following packages provided they would prevent loss or dispersal of their radioactive contents with a reduction of ambient pressure to 25kPa:

(a) packages containing only special form radioactive material, SCO-I or SCO-II;

(b) excepted packages, industrial packages and type A packages, containing solid radioactive material, excluding powders;

(c) packages containing LSA-I material in powder form.

This proposal was completed by the following recommendations to be introduced into the IAEA guide SSG-26:

621.1. This is a similar provision to that required by the ICAO [10] for packages containing liquid dangerous goods or infectious substances intended for transport by air. The basic requirement is that the package shall be capable of withstanding, without loss or dispersal of radioactive contents from the containment system, an internal pressure that produces a pressure differential of not less than the maximal gauge pressure that may be developed in the package at 55°C when it has been filled at 15°C (or the maximum normal operating pressure - MNOP) plus 75kPa, with a minimum of 95kPa. The pressure of 75kPa corresponds to the highest pressure differential between sea level and the pressure within a non-pressurized or partially pressurized cargo compartment of an airplane (see note 3 in introductory notes of Part 4 of ICAO-TI 2015-2016 Edition). An alternative requirement is proposed for the cases where dispersal of the radioactive material is improbable due to its form (special form, SCO, solid LSA in non-powder form, solid radioactive material in a Type A), and for LSA-I in powder form, in these cases it is required that the package shall retain its radioactive contents under a reduction of ambient pressure to 25kPa. This pressure corresponds to the pressure at sea level (100kPa) less 75kPa. It also corresponds to the pressure existing at an altitude of about 10350 m above sea level. The minimum differential pressure of 95kPa is the same as the one specified for solid infectious substances (UN2814, UN2900 and UN3373) and liquid dangerous goods from other classes.

621.2 Several possibilities to comply with the requirement may be used:

- In the case of solid radioactive material in non-approved packages, for which, except the increase of pressure of the content gases (air or other gas) due to heating of the package from 15 to 55°C, no other source of increase of the pressure is expected (low activities, no heating by the radioactive content, no production of gases by radiolysis phenomenon),
packaging able to withstand a pressure differential of 95kPa comply to the requirement: the differential pressure of 95kPa covers the cumulative effect of the highest pressure differential between sea level and the pressure within a non-pressurized or partially pressurized cargo compartment of an airplane (75kPa) and the increase of air pressure due to increase of temperature between 15°C and 55°C (15kPa).

- In the case of use of the alternative requirements, it is possible to use a packaging equipped with valves and filters to allow pressure equilibrium without dispersal of radioactive solid material content when the content can be subject to dispersal like LSA-I, or small dispersal like SCO-I and SCO-II, or to use a packaging equipped with valves without any supplementary equipment if the content is not subject to dispersal, like special form.

- However, the characteristics of the filter must be such that the pressure differential due to the presence of the filter during the phases of depressurization and pressurization of the aircraft in normal flight conditions does not exceed the capacity of containment of the package. It can be assumed, in routine and normal conditions of transport, that the variation of the pressure does not exceed 30Pa/s (1.8kPa/min corresponding to a cabin altitude rate of 500ft/min in normal flight conditions) inside a pressurized aircraft or 150 Pa/s (9kPa/min corresponding to a cabin altitude rate of 2500ft/min in normal flight conditions) inside a non-pressurized or partially pressurized aircraft. When filters are needed, they should be mounted in series with each of the valves or equivalent devices used to equalize the pressure in normal flight conditions.

- In addition, packages and containers having internal volume greater than 1 m³, for which those alternative requirements are used, should be designed to be able to ensure high flow pressure equalization in the event of a rapid decompression at cruise altitude without creating a hazard to the cargo compartment or aircraft structure. A rapid decompression is a situation, due to an incident or accident, where the indoor ambient pressure of an airplane pressurized at its cruise altitude drops suddenly to the external pressure at the altitude of the aircraft flight. Rapid decompression is neither routine nor normal condition of transport as defined in para. 106 of SSR-6. See ISO 11242:1996 standard (Aircraft – Pressure equalization requirements for cargo containers) for detailed conditions and requirements.
Appendix 2

Extracts from US and EU certification specifications for large aeroplanes

US FAA – FARs – Part 25 – Section 841 – Pressurized cabins:

(a) Pressurized cabins and compartments to be occupied must be equipped to provide a cabin pressure altitude of not more than 8,000 feet at the maximum operating altitude of the airplane under normal operating conditions.

   (1) If certification for operation above 25,000 feet is requested, the airplane must be designed so that occupants will not be exposed to cabin pressure altitudes in excess of 15,000 feet after any probable failure condition in the pressurization system.

   (2) The airplane must be designed so that occupants will not be exposed to a cabin pressure altitude that exceeds the following after decompression from any failure condition not shown to be extremely improbable:

      (i) Twenty-five thousand (25,000) feet for more than 2 minutes; or

      (ii) Forty thousand (40,000) feet for any duration.

   (3) Fuselage structure, engine and system failures are to be considered in evaluating the cabin decompression.

(b) Pressurized cabins must have at least the following valves, controls, and indicators for controlling cabin pressure:

   (1) Two pressure relief valves to automatically limit the positive pressure differential to a predetermined value at the maximum rate of flow delivered by the pressure source. The combined capacity of the relief valves must be large enough so that the failure of any one valve would not cause an appreciable rise in the pressure differential. The pressure differential is positive when the internal pressure is greater than the external.

   (2) Two reverse pressure differential relief valves (or their equivalents) to automatically prevent a negative pressure differential that would damage the structure. One valve is enough, however, if it is of a design that reasonably precludes its malfunctioning.

   (3) A means by which the pressure differential can be rapidly equalized.

   (4) An automatic or manual regulator for controlling the intake or exhaust airflow, or both, for maintaining the required internal pressures and airflow rates.

   (5) Instruments at the pilot or flight engineer station to show the pressure differential, the cabin pressure altitude, and the rate of change of the cabin pressure altitude.

   (6) Warning indication at the pilot or flight engineer station to indicate when the safe or preset pressure differential and cabin pressure altitude limits are exceeded. Appropriate warning markings on the cabin pressure differential indicator meet the warning requirement for pressure differential limits and an aural or visual signal (in addition to cabin altitude indicating means) meets the warning requirement for cabin pressure altitude limits if it warns the flight crew when the cabin pressure altitude exceeds 10,000 feet.

   (7) A warning placard at the pilot or flight engineer station if the structure is not designed for pressure differentials up to the maximum relief valve setting in combination with landing loads.
(8) The pressure sensors necessary to meet the requirements of paragraphs (b)(5) and (b)(6) of this section and §25.1447(c), must be located and the sensing system designed so that, in the event of loss of cabin pressure in any passenger or crew compartment (including upper and lower lobe galleys), the warning and automatic presentation devices, required by those provisions, will be actuated without any delay that would significantly increase the hazards resulting from decompression.

EU - EASA CS25 - CS 25.841 - Pressurised cabins

(a) Pressurised cabins and compartments to be occupied must be equipped to provide a cabin pressure altitude of not more than 2438 m (8000 ft) at the maximum operating altitude of the aeroplane under normal operating conditions. If certification for operation over 7620 m (25 000 ft) is requested, the aeroplane must be able to maintain a cabin pressure altitude of not more than 4572 m (15 000 ft) in the event of any reasonably probable failure or malfunction in the pressurisation system.

(b) Pressurised cabins must have at least the following valves, controls, and indicators for controlling cabin pressure:

1. Two pressure relief valves to automatically limit the positive pressure differential to a predetermined value at the maximum rate of flow delivered by the pressure source. The combined capacity of the relief valves must be large enough so that the failure of any one valve would not cause an appreciable rise in the pressure differential. The pressure differential is positive when the internal pressure is greater than the external.

2. Two reverse pressure differential relief valves (or their equivalents) to automatically prevent a negative pressure differential that would damage the structure. One valve is enough, however, if it is of a design that reasonably precludes it's malfunctioning.

3. A means by which the pressure differential can be rapidly equalised.

4. An automatic or manual regulator for controlling the intake or exhaust airflow, or both, for maintaining the required internal pressures and airflow rates.

5. Instruments at the pilot or flight engineer station to show the pressure differential, the cabin pressure altitude, and the rate of change of the cabin pressure altitude.

6. Warning indication at the pilot or flight engineer station to indicate when the safe or preset pressure differential and cabin pressure altitude limits are exceeded. Appropriate warning markings on the cabin pressure differential indicator meet the warning requirement for pressure differential limits and an aural or visual signal (in addition to cabin altitude indicating means) meets the warning requirement for cabin pressure altitude limits if it warns the flight crew when the cabin pressure altitude exceeds 3048 m (10 000 ft).

7. A warning placard at the pilot or flight engineer station if the structure is not designed for pressure differentials up to the maximum relief valve setting in combination with landing loads.

8. The pressure sensors necessary to meet the requirements of sub-paragraphs (b)(5) and (b)(6) of this paragraph and CS 25.1447 (c), must be located and the sensing system designed so that, in the event of loss of cabin pressure in any passenger or crew compartment (including upper and lower lobe galleys), the warning and automatic presentation devices, required by those provisions, will be actuated without any delay that would significantly increase the hazards resulting from decompression.
Appendix 3

Proposed amendment to para 621 of the IAEA SSR-6 and to para 621.1 and following of IAEA SSG-26 with account taken of depressurization incidents as representing NCT

621. Packages containing radioactive material to be transported by air shall be capable of withstanding, without loss or dispersal of radioactive contents from the containment system, an internal pressure that produces a pressure differential of not less than the maximal gauge pressure that may be developed in the package at 55°C when it has been filled at 15°C (or the maximum normal operating pressure) plus 75kPa, with a minimum of 95kPa. This requirement is not applicable to the following packages provided they would prevent loss or dispersal of their radioactive contents with a reduction of ambient pressure from 100 kPa to 25 kPa at rate of 9 kPa/minute in routine conditions of transport, and from 75 kPa to 25 kPa at rate of 25 kPa/minute in normal conditions of transport when applicable:

(a) packages containing only special form radioactive material, SCO-I or SCO-II;
(b) excepted packages, industrial packages and type A packages, containing solid radioactive material, excluding powders;
(c) packages containing LSA-I material in powder form.

This proposal is completed by the following recommendations to be introduced into the IAEA guide SSG-26:

621.1. This is a similar provision to that required by the ICAO [10] for packages containing liquid dangerous goods or infectious substances intended for transport by air. The basic requirement is that the package shall be capable of withstanding, without loss or dispersal of radioactive contents from the containment system, an internal pressure that produces a pressure differential of not less than the maximal gauge pressure that may be developed in the package at 55°C when it has been filled at 15°C (or the maximum normal operating pressure - MNOP) plus 75kPa, with a minimum of 95kPa. The pressure of 75kPa corresponds to the highest pressure differential between sea level and the pressure within a non-pressurized or partially pressurized cargo compartment of an airplane (see note 3 in introductory notes of Part 4 of ICAO -TI 2015-2016 Edition). An alternative requirement is proposed for the cases where dispersal of the radioactive material is improbable due to its form (special form, SCO, solid LSA in non-powder form, solid radioactive material in a Type A), and for LSA-I in powder form, in these cases it is required that the package shall retain its radioactive contents under a reduction of ambient pressure to 25kPa. This pressure corresponds to the pressure at sea level (100kPa) less 75kPa. It also corresponds to the pressure existing at an altitude of about 10350 m above sea level. The minimum differential pressure of 95kPa is the same as the one specified for solid infectious substances (UN2814, UN2900 and UN3373) and liquid dangerous goods from other classes.

621.2 Several possibilities to comply with the requirement may be used:

• In the case of solid radioactive material in non-approved packages, for which, except the increase of pressure of the content gases (air or other gas) due to heating of the package from 15 to 55°C, no other source of increase of the pressure is expected (low activities, no heating by the radioactive content, no production of gases by radiolysis phenomenon), packaging able to withstand a pressure differential of 95kPa comply to the requirement: the
differential pressure of 95kPa covers the cumulative effect of the highest pressure
differential between sea level and the pressure within a non-pressurized or partially
pressurized cargo compartment of an airplane (75kPa) and the increase of air pressure
due to increase of temperature between 15°C and 55°C (14kPa).

• In the case of use of the alternative requirements, it is possible to use a packaging equipped
with valves and filters to allow pressure equilibrium without dispersal of radioactive solid
material content when the content can be subject to dispersal like LSA-I, or small dispersal
like SCO-I and SCO-II, or to use a packaging equipped with valves without any
supplementary equipment if the content is not subject to dispersal, like special form.

• However, the characteristics of the filter must be such that the pressure differential due to
the presence of the filter during the phases of depressurization and pressurization of the
aircraft in normal flight conditions does not exceed the capacity of containment of the
package. It can be assumed, in routine and normal conditions of transport, that the variation
of the pressure does not exceed 30Pa/s (1.8kPa/min corresponding to a cabin altitude rate
of 500ft/min in normal flight conditions) inside a pressurized aircraft or 150 Pa/s (9kPa/min
corresponding to a cabin altitude rate of 2500ft/min in normal flight conditions) inside a non-
pressurized or partially pressurized aircraft. When filters are needed, they should be
mounted in series with each of the valves or equivalent devices used to equalize the
pressure in normal flight conditions.

• In addition, packages and containers having internal volume greater than 1 m³, for which
those alternative requirements are used, should be designed to be able to ensure high flow
pressure equalization in the event of a rapid decompression at cruise altitude without
creating a hazard to the cargo compartment or aircraft structure. A rapid decompression is
a situation, due to an incident or accident, where the indoor ambient pressure of an airplane
pressurized at its cruise altitude drops suddenly to the external pressure at the altitude of
the aircraft flight. Rapid decompression is neither routine nor is considered as part of the
normal condition of transport as defined in para. 106 of SSR-6. The rate of 25kPa/minute
corresponds to an increase in cabin altitude of 26,000 ft in two minutes (increase of the
cabin altitude from 8,000 ft to 34,000 ft in 2 minutes which corresponds drop of the ambient
pressure from 75 kPa to 25 kPa in 2 minutes). See ISO 11242:1996 standard (Aircraft –
Pressure equalization requirements for cargo containers) for detailed conditions and
requirements. Packagings equipped with valves for pressure equilibrium under a rate of 9
kPa/minute which in addition are able to withstand a pressure differential of at least 50 kPa
does not need to be equipped with emergency valves to equalize the pressures in the event
of a rapid decompression.